

## Finite element implementation for the analysis of 3D steel and composite frames subjected to fire

### Abstract

This paper presents a finite element formulation for the numerical analysis of three-dimensional framed steel, reinforced concrete or composite steel and concrete structures subjected to fire. Several specialized and commercial programs may be used for the analysis of structures in fire condition. Within this context, the purpose of this work is to present the steps taken to extend a previously developed static analysis procedure with beam elements in order to cope with the thermal and structural analysis of structures under fire action. Physical nonlinearity and material property degradation considering the temperature distribution are taken into account at the cross section level, which is divided into quadrilateral or triangular finite elements. Thermal strains are considered through the effective strain concept, and the resulting nonlinear system of equations is solved by the Newton-Raphson scheme. The accuracy and capability of the formulation to simulate the behavior of framed structures under fire action are assessed through comparison with various numerical and experimental results.

### Keywords

structures, fire, finite element, steel, concrete, composite steel and concrete

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## 1 INTRODUCTION

Research on the behavior of structures under fire conditions started in the late 19th century, motivated by losses of buildings due to structural collapse caused by fires. Since then, fire engineering research has grown steadily. In recent decades, experimental tests have provided fundamental information on structures under fire action and advances on computational resources have led to the development of numerical procedures capable of simulating their behavior. Due to the high costs associated with full-scale fire tests, however, most of the experimental work has been focused on isolated elements. Some important effects such as load redistribution and the ability to perform parametric studies are therefore more suitable in a computational environment. Nowadays, numerical methods are used either for the design of complex structures in real projects or for the develop-

ment and verification of design methods. A brief review of the main computational developments in the field is presented below.

According to Franssen (2005), some of the first numerical models for fire analysis were developed at the University of California, Berkeley, where programs such as FIRES-T and FIRES-RC (Becker, et al., 1974; Becker, et al., 1974) were employed in the analysis of reinforced concrete elements.

The VULCAN program was developed by researchers at the University of Sheffield, UK (Saab and Nethercot, 1991; Najjar, et al., 1996). Huang et al. (2003) implemented an isoparametric three-noded (6 d.o.f per node) beam element under a total lagrangian formulation and a second-order approximation for the rotation tensor (Bathe, 1996). The cross section is divided into a matrix of segments which allows for the simulation of generic reinforced concrete, steel or composite sections (Cai, et al., 2003; Huang, et al., 2009). The program presents semi-rigid connection modeling (Bailey, 1995; Block, et al., 2006) and shell elements for the analysis of reinforced concrete slabs (Huang, et al., 1999). VULCAN has been employed in a large number of works (Bailey, et al., 1999; Burgess, et al., 1998; Huang, et al., 2006; Abu, et al., 2010).

The SAFIR program (Franssen, 1987) was developed at Liège University, Belgium, and has been used for extensive research on fire behavior of concrete, steel and composite structures as well as in the assessment of design codes (Vila Real, et al., 2004; Franssen, et al., 2006). The program can perform two and three-dimensional thermal analysis, and for structural analysis, three-dimensional beam, shell and solid elements are available. The three-dimensional beam element is based on a corotational formulation, with seven degrees of freedom at the end nodes (warping included) and a central node which interpolates the nonlinear part of the axial strain (Franssen, 2005; Franssen, 1997).

Commercial FE programs such as ABAQUS and ANSYS have also been used for research on analysis of structural behavior under fire conditions. They are capable of taking into account the change in material properties with temperature and performing heat transfer analysis (Gillie, et al., 2001; O'Connor, et al., 1998; Li, et al., 2006).

Most of the cited works with beam-columns was based on some form of distributed plasticity approach, with a matrix of segments at the cross section level. Other formulations that adopt the plastic hinge approach have also been successfully implemented (Ma, et al., 2004; Iu, et al., 2005; Landesmann, et al., 2005; Souza Jr., et al., 2007).

The purpose of this work is to present a three-dimensional beam element for the structural and thermal analysis of reinforced concrete, steel and composite steel and concrete framed structures and its application to fire analysis (Caldas, 2008). The FE model is based on the beam-column element used to analyze composite columns by Sousa Jr. and Caldas (Sousa Jr. and Caldas, 2005) and the well-known corotational technique (Crisfield, 1997). Physical nonlinearity is taken into account at the cross section level divided into a matrix of segments which considers material property degradation under temperature action. Thermal strains are considered by means of the effective strain concept (Parente Jr., et al., 2006) and the resulting nonlinear system of equations is solved by the Newton-Raphson scheme. At the cross section level, each segment is a quadrilateral or triangular finite element. The mesh of elements is used to evaluate the temperature distribution in a generic composite section. Numerical examples presented at the end of the paper show the applica-

bility of the formulation for plain steel, reinforced concrete and composite steel and concrete framed structures.

## 2 NUMERICAL MODEL

The numerical model developed here for the fire analysis of 3D framed structures is based on a thermomechanical analysis which is characterized by the influence of temperature elevation on the strain state and material property degradation under fire (Caldas, 2008). The effects of temperature variation are taken into account at the cross section (integration point) level. The finite element is a displacement-based generic cross section beam-column element (Sousa Jr. and Caldas, 2005) in conjunction with the corotational formulation, which accounts for large displacements and rotations (Caldas, 2008).

### 2.1 Cross section thermomechanical analysis

Section geometry is represented by quadrangular or triangular linear finite element mesh. A uniaxial stress state is considered for the mechanical evaluations, which employ the effective strain concept, by which  $\boldsymbol{\varepsilon}_{ef} = \boldsymbol{\varepsilon}_x - \boldsymbol{\varepsilon}_{th}$ , where  $\boldsymbol{\varepsilon}_x$  is the total strain and  $\boldsymbol{\varepsilon}_{th}$  are the thermal strains evaluated at each element centroid. Strains are evaluated by  $\boldsymbol{\varepsilon}_x = \boldsymbol{\varepsilon}_0 + k_y z - k_z y$ , where  $\boldsymbol{\varepsilon}_0$  is the strain at a reference point (usually the cross section centroid) and  $k_y, k_z$  are the curvatures of the section, assumed to lie in the  $y-z$  plane. The resistant forces of the cross section are obtained by numerical integration using the finite elements  $i = 1 \dots n$ , with area  $A$ :

$$N_x = \sum_{i=1}^n (\sigma_x A)_i, \quad M_y = \sum_{i=1}^n (\sigma_x z A)_i \quad e \quad M_z = -\sum_{i=1}^n (\sigma_x y A)_i. \quad (1)$$

The tangent moduli (derivatives of the forces with respect to the deformational variables) are obtained in a similar fashion. The heat transfer analysis is performed employing an explicit time integration scheme (Cook, et al., 2002). In order to ensure stability, the vector of temperature increments  $\Delta \mathbf{a}$  in the elements of the cross section is evaluated for a limited time increment  $\Delta t$  as

$$\Delta \mathbf{a} = \mathbf{C}^{-1} (\mathbf{F} - \mathbf{K} \mathbf{a}) \Delta t. \quad (2)$$

where  $\mathbf{K}$  is the conductivity matrix,  $\mathbf{F}$  is the boundary heat flux vector and  $\mathbf{C}$  is the specific heat matrix. In the implementation, concrete, steel and fire protection material properties are considered. For the concrete, spalling is not taken into account, and the presence of moisture is considered by modifying the thermal properties as specified by Eurocode 2 (2004). The uniaxial stress-strain relationships for concrete with strength reduction, strain at peak stress and ultimate strain are considered as specified by the Eurocode 2 (2004) for siliceous or calcareous aggregates. Figure 1a gives an example for concretes with siliceous aggregates. The uniaxial

stress-strain relationship for steel is considered as specified by the Eurocode 3 (2005), Figure 1b.

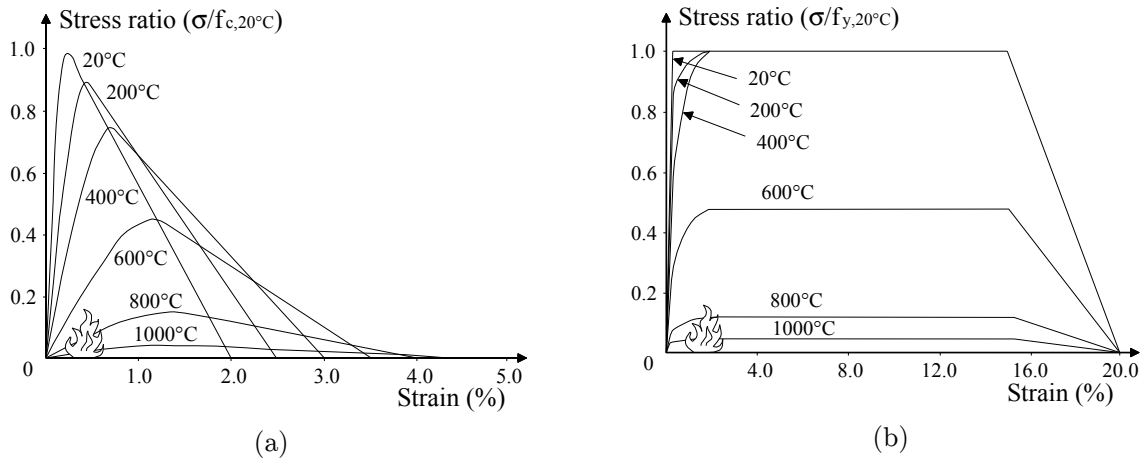


Figure 1 Stress-strain relationships of concrete in compression and steel at elevated temperature: (a) concrete in compression; (b) steel

Concrete heating to a maximum temperature and subsequently cooling down does not recover its initial compressive strength. The stress-strain relationships adapted to natural fires with a decreasing heating branch presented in the informative annex C of the Eurocode 4 (2005) are considered. The concrete tensile stress is considered as in Huang et al. (2003) (Figure 2), with  $\epsilon_{cu} = 15\epsilon_{cr}$ . The tensile strength reduction is considered equal to the cubic power of the compressive strength reduction.

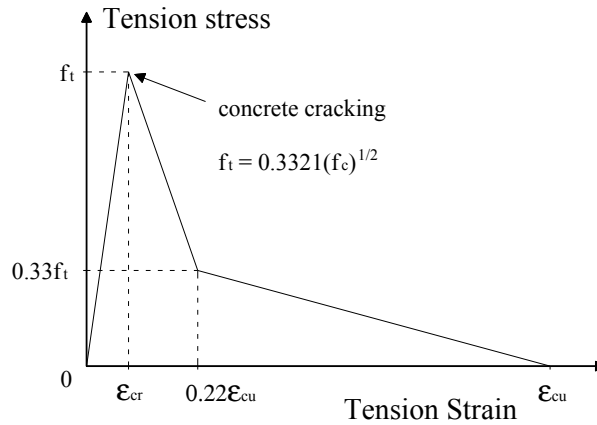


Figure 2 Stress-strain relationships of concrete in tension (Huang, et al., 2003)

### 2.2 Finite element formulation

The finite element employed in the present work must be able to undergo large displacements and rotations, but small strains. Basic assumptions include: plane sections remain plane; there is perfect bond at the interface between materials; each element is subjected to a uniaxial state of stress; local buckling and cross section warping are not taken into account.

The corotational (CR) formulation is employed for the treatment of large displacements and rotations. Roughly speaking, the CR formulation consists of a set of axes which continuously moves and rotates with the element. Relative to this moving frame a “local” element formulation is defined, and large displacement effects are taken into account in the transformation between the local and global coordinate systems. In two-dimensional problems the transformations between the systems are exact, but in three-dimensional problems different formulations may be derived based on different definitions of the local system and assumptions relative to the global and local rotations. The present work follows the procedure developed by Crisfield (1997). A good explanation and detailed derivation is also presented by Souza (2000). The main geometric operations are briefly described below.

Figure 3 displays the straight undeformed element and its initial local system  $\hat{x}\hat{y}\hat{z}$ , correspondent to a triad of three base vectors (axes)  $\hat{e}_1$ ,  $\hat{e}_2$  and  $\hat{e}_3$ , defined for the two nodes and which may be collected as a rotation matrix

$$\hat{\mathbf{E}} = \begin{bmatrix} \hat{e}_1 & \hat{e}_2 & \hat{e}_3 \end{bmatrix}. \tag{3}$$

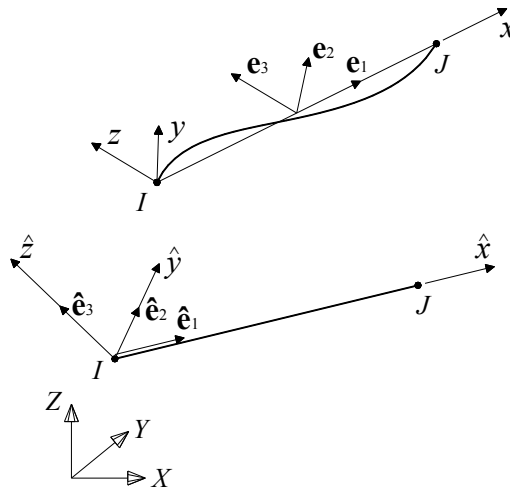


Figure 3 Initial element spatial configuration

During deformation the two element nodal triads of axes will translate and rotate in space, and in the current deformed configuration, the new positions for the triads (Figure 4) are written as

$$\mathbf{N}_I = \begin{bmatrix} \mathbf{n}_{I1} & \mathbf{n}_{I2} & \mathbf{n}_{I3} \end{bmatrix} \quad \text{and} \quad \mathbf{N}_J = \begin{bmatrix} \mathbf{n}_{J1} & \mathbf{n}_{J2} & \mathbf{n}_{J3} \end{bmatrix}, \quad (4)$$

which in the initial position coincide with matrix  $\hat{\mathbf{E}}$ .

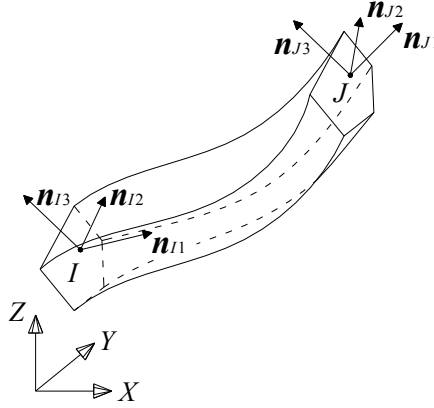


Figure 4 Nodal triads of axes in deformed configuration

The position of the triads of axes is updated by means of incremental translation and rotation values. The rotation increments, which cannot be simply added due to the non-commutative property of rotations in space, allow one to update the rotational variables. The rotation variables for each node may be a pseudovector, its associated rotation matrix, or a unit quaternion (Crisfield, 1997).

After element deformation, the element-attached coordinate system  $xyz$ , used to evaluate the “local” element displacements and strains, may be defined as follows: the  $x$  axis, which corresponds to unit vector  $\mathbf{e}_1$ , is defined by the direction of the line connecting the end nodes in the deformed configuration, and the other two directions are defined by unit vectors  $\mathbf{e}_2$  and  $\mathbf{e}_3$  (Figure 3), given by

$$\mathbf{e}_2 = \bar{\mathbf{r}}_2 - \frac{\bar{\mathbf{r}}_2^T \mathbf{e}_1}{2} (\bar{\mathbf{r}}_1 + \mathbf{e}_1) \quad \text{and} \quad \mathbf{e}_3 = \bar{\mathbf{r}}_3 - \frac{\bar{\mathbf{r}}_3^T \mathbf{e}_1}{2} (\bar{\mathbf{r}}_1 + \mathbf{e}_1), \quad (5)$$

where the vectors  $\bar{\mathbf{r}}_1$ ,  $\bar{\mathbf{r}}_2$  and  $\bar{\mathbf{r}}_3$  compose the triad

$$\bar{\mathbf{R}} = \begin{bmatrix} \bar{\mathbf{r}}_1 & \bar{\mathbf{r}}_2 & \bar{\mathbf{r}}_3 \end{bmatrix}. \quad (6)$$

The definition of this local system is not unique and the solution presented above was originally suggested by Crisfield (1997). The triad  $\bar{\mathbf{R}}$  corresponds to an averaged rotation between

the two nodal rotations. “Local” element rotations (element degrees of freedom) maybe then recognized (Figure 5) as

$$\begin{aligned}
 \theta_{xI} &= \arcsin \left[ \frac{1}{2} (\mathbf{e}_3^T \mathbf{n}_{I2} - \mathbf{e}_2^T \mathbf{n}_{I3}) \right], & \theta_{yI} &= \arcsin \left[ \frac{1}{2} (\mathbf{e}_1^T \mathbf{n}_{I3} - \mathbf{e}_3^T \mathbf{n}_{I1}) \right], \\
 \theta_{xJ} &= \arcsin \left[ \frac{1}{2} (\mathbf{e}_2^T \mathbf{n}_{J1} - \mathbf{e}_1^T \mathbf{n}_{J2}) \right], & \theta_{yJ} &= \arcsin \left[ \frac{1}{2} (\mathbf{e}_3^T \mathbf{n}_{J2} - \mathbf{e}_2^T \mathbf{n}_{J3}) \right], \\
 \theta_{zI} &= \arcsin \left[ \frac{1}{2} (\mathbf{e}_1^T \mathbf{n}_{J3} - \mathbf{e}_3^T \mathbf{n}_{J1}) \right] & \theta_{zJ} &= \arcsin \left[ \frac{1}{2} (\mathbf{e}_2^T \mathbf{n}_{J1} - \mathbf{e}_1^T \mathbf{n}_{J2}) \right].
 \end{aligned}
 \tag{7}$$

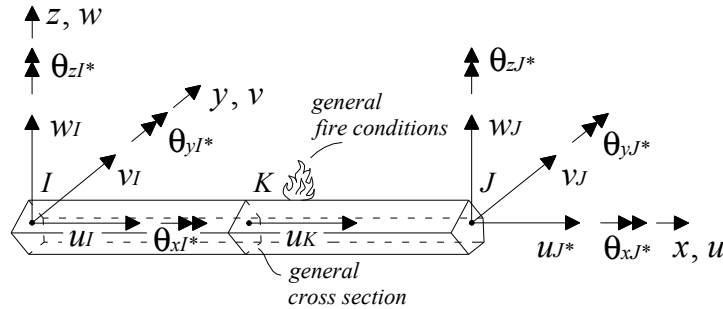


Figure 5 Local system (CR formulation d.o.f.'s are indicated by the symbol “\*”)

It is assumed that the local element has seven degrees of freedom, six rotations evaluated from Equations 7 and the axial displacement  $u_j$  in the direction  $\mathbf{e}_1$  (Figures 3 and 5). The incremental relation between displacements in the global ( $\delta \mathbf{u}$ ) and local ( $\delta \hat{\mathbf{u}}$ ) systems is given by

$$\delta \mathbf{u} = \frac{\partial \mathbf{u}}{\partial \hat{\mathbf{u}}} \delta \hat{\mathbf{u}} = \mathbf{T} \delta \hat{\mathbf{u}}.
 \tag{8}$$

Matrix  $\mathbf{T}$  is obtained by geometrical considerations, involving considerable algebraic manipulation, which may be found in Crisfield (1997) or Souza (2000). Its transpose gives the relationship between forces in the two systems

$$\hat{\mathbf{P}} = \mathbf{T}^T \mathbf{P}.
 \tag{9}$$

The local stiffness matrix relates displacement and force increments

$$\delta \mathbf{P} = \mathbf{K} \delta \mathbf{u}.
 \tag{10}$$

Through linearization of Expression (9) with Expressions (8) and (10) one may obtain the expression for the tangent stiffness in the global coordinate system

$$\delta \hat{\mathbf{P}} = (\mathbf{T}^T \mathbf{K} \mathbf{T} + \mathbf{K}_g) \delta \hat{\mathbf{u}} = \hat{\mathbf{K}} \delta \hat{\mathbf{u}}, \quad (11)$$

where the geometric stiffness  $\mathbf{K}_g$  is given by

$$\mathbf{K}_g = \frac{\partial \mathbf{T}^T}{\partial \hat{\mathbf{u}}} : \mathbf{P}, \quad (12)$$

and where “:” represents a contraction, such that

$$\delta \mathbf{T}^T \mathbf{P} = \sum_{r=1}^3 \delta \mathbf{t}_r P_r = \mathbf{K}_g \delta \hat{\mathbf{u}}, \quad (13)$$

and  $\mathbf{t}_r$  are the lines of transformation matrix  $\mathbf{T}$ . To obtain the geometric stiffness matrix the variation of  $\mathbf{T}$  is required, which once again leads to a large amount of algebraic manipulation.

The geometric stiffness matrix is non-symmetric. It has been observed, however, that artificial symmetrization does not lead to severe convergence problems. Upon reaching equilibrium, however, symmetry is recovered.

In the Newton-Raphson (N-R) scheme for fire analysis, load increments are applied prior to time increments of fire exposure. After applying a typical time increment, a heat transfer analysis is performed on each integration point. With the new thermal strains and material properties, the internal force is no longer in equilibrium with the applied loads, and an iterative N-R scheme is carried out to recover equilibrium. Structural collapse is identified when the numerical scheme is not able to reach an equilibrium point.

### 2.3 Local element formulation

In the expression of the CR element stiffness formulation matrix the only requirement on the “local” element formulation is to have the same seven degrees of freedom shown in Figure 5 (indicated with the symbol “\*”). The 11 degree of freedom finite element employed by Sousa Jr. and Caldas (2005) for the numerical analysis of composite columns will be used, with uncoupled torsional stiffness added. The element is based on a total lagrangian formulation, suitable for large displacements and moderate rotations. Cubic hermitian functions are used for local transverse displacement interpolations,  $v = \mathbf{N}_v^T \mathbf{q}_v$  and  $w = \mathbf{N}_w^T \mathbf{q}_w$ . For the axial displacements, quadratic functions are employed, with a hierarchical axial displacement degree of freedom (d.o.f.),  $u = \mathbf{N}_u^T \mathbf{q}_u$ . The additional d.o.f. is dealt with by means of static condensation at the



incremental level. This element performed very well for nonlinear mechanical analysis of composite beam-columns (Sousa Jr. and Caldas, 2005).

The Principle of Virtual Work, associated with the strain-displacement relation

$$\epsilon_x = u' + 1/2[(v')^2 + (w')^2] - yv'' - zw'' \tag{14}$$

where the prime indicates differentiation with respect to  $x$ , allows one to obtain the internal force vector

$$\mathbf{p}_m = \int_{\ell_m} \begin{bmatrix} N_x \mathbf{N}'_u \\ N_x v' \mathbf{N}'_v + M_z \mathbf{N}''_v \\ N_x w' \mathbf{N}'_w - M_y \mathbf{N}''_w \end{bmatrix} dx, \tag{15}$$

and the local element tangent stiffness

$$\mathbf{k}_m = \int_{\ell_m} \begin{bmatrix} \varphi'_u \left\{ \frac{\partial N_x}{\partial \mathbf{q}} \right\}^T \\ \mathbf{N}'_v \left( v' \left\{ \frac{\partial N_x}{\partial \mathbf{q}} \right\}^T + N_x \begin{bmatrix} \mathbf{0}_u & \mathbf{N}'_v & \mathbf{0}_w \end{bmatrix} \right) + \mathbf{N}''_v \left\{ \frac{\partial M_z}{\partial \mathbf{q}} \right\}^T \\ \mathbf{N}'_w \left( w' \left\{ \frac{\partial N_x}{\partial \mathbf{q}} \right\}^T + N_x \begin{bmatrix} \mathbf{0}_u & \mathbf{0}_v & \mathbf{N}'_w \end{bmatrix} \right) - \mathbf{N}''_w \left\{ \frac{\partial M_y}{\partial \mathbf{q}} \right\}^T \end{bmatrix} dx, \tag{16}$$

where the section resultant forces  $N_x$ ,  $M_y$ ,  $M_z$  as well as their derivatives with respect to the nodal displacements  $\mathbf{q}$  are obtained by employing finite element mesh at cross section level.

The influence of torsion and its corresponding degree of freedom are taken into account by the inclusion of a linear relation  $\theta = T \ell_o / GJ$ , where the user-defined term  $GJ$  may take into account the effect of elevated temperature and/or cracking.

Warping may be included in the “local” element formulation (Franssen, 2005; Najjar, et al., 1996; Bailey, 1995), and in this case the geometric stiffness would be unchanged (Battini, et al., 2002). Nonetheless, in most cases the influence of warping in the global structural behavior of frames may be neglected.

By departing from the element with 13 degrees of freedom (6 at each node and the hierarchic axial displacement) and suppressing some lines and columns using the static condensation procedure, an element which has seven desired degrees of freedom is obtained, Figure 5.

## 2.4 Program implementation

The steps taken to extend the previously developed scheme by Sousa Jr. and Caldas (2005) can be summarized as follows:

- For consideration of fire action, the main modifications occur at the cross section level and include thermal analysis, considerations of degradation of materials properties in the stress-strain relationships and thermal strains. Alternatively, one may perform thermal analysis in an independent program or previous step and use the temperature distribution as data for the structural analysis;
- The CR formulation allows one to analyze structures with large displacements and rotations. This implementation step is not necessary in a static analysis procedure with these capacities;
- At global level, the Newton-Raphson scheme is adapted to apply time increments of fire exposure after load increments.

As the beam-column element is implemented in a general purpose finite element program, with object-oriented techniques employing the C++ language, the changes to the previous scheme (Sousa Jr. and Caldas, 2005) include: the implementation of a method (class) for consideration of adaptations at cross section level (for instance, thermal analysis); modifications to the method of calculation of the stiffness matrix and internal forces vector in order to take the CR formulation into account; and, another method for the Newton-Raphson scheme with time increments of fire exposure.

## 3 EXAMPLES

The beam-column finite element developed in the present work was tested against various experimental and numerical results. The numerical values for comparison come from results obtained from other works which used the VULCAN and SAFIR programs. Some results for VULCAN were generated with a version of the program available at Federal University of Minas Gerais (version 10.0). In all of the examples, the present formulation employed temperature-dependent material properties from Eurocode 2 (2004) or Eurocode 3 (2005).

### 3.1 Steel column

A pin-ended IPE360 steel column, 4m in length, subjected to the standard ISO 834-1 time-temperature curve with fire exposure on three and four faces was analyzed by Landesmann et al. (2005), employing SAFIR, as well as by the authors of this work, with VULCAN version 10.0 and with the present formulation. At ambient temperature the column is loaded by an axial force equal to 30% of the full plastic strength of the section, and by a constant moment (about the stronger axis and causing single curvature) which is 20% of the section's full plastic moment.

In the present work four equal sized elements were employed, with four gauss points each. For the column with three faces exposed, the face not exposed to fire was modeled as being protected by adiabatic material. The behavior obtained with the present formulation (Figure 6) is almost identical to the results of SAFIR (Franssen, 2005) and similar to the results obtained with VULCAN by the authors.

In the analysis with VULCAN, the cross section temperatures used as input were taken as being equal to a reference temperature obtained at the centroid of the unprotected flange. For the columns exposed on three faces, the temperatures at the web and the protected flange were taken as 100% and 80% of this reference respectively. This distribution was based on thermal analysis of the cross section. It was found, however, that this choice of distribution does not have significant influence on the results obtained.

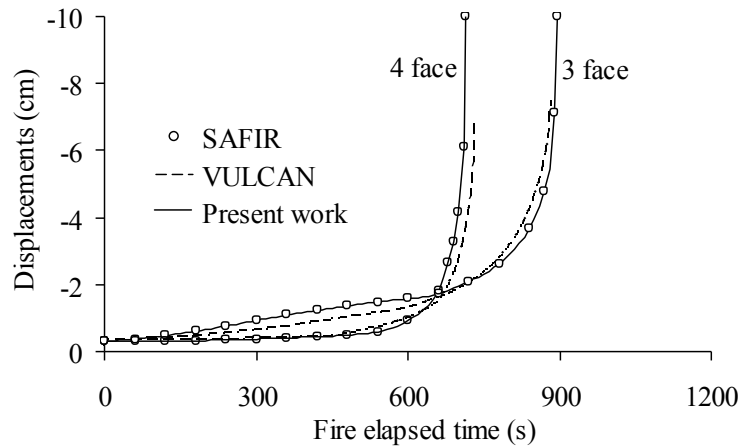


Figure 6 Results for steel columns

### 3.2 Steel beam

Landesmann et al. (2005) also employed SAFIR to analyze the fire behavior of a simply supported beam, 5m in length, subjected to end moments which vary from 0.1 to 0.9 times the ambient temperature full plastic moment of the IPE360 cross section. For the present formulation four equal elements with four integration points were employed. The top flange's upper face was considered protected by a material with adiabatic thermal properties.

In the analysis with VULCAN 10.0 the temperature of the protected flange was taken as being 80% of the unprotected flange for the beams with an applied moment between 0.7 and 0.9 times the plastic moment, and as being 90% in the other cases. These values were based on thermal analyses of the IPE360 profile.

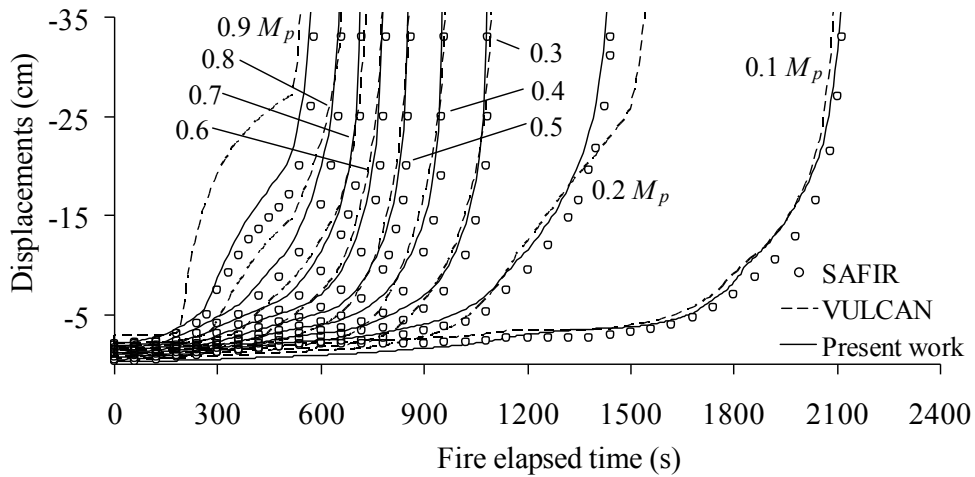


Figure 7 Results for steel beams under constant moment

Figure 7 displays the results for the mid-span displacements. The beam is initially loaded at ambient temperature and then the time is increased to simulate fire exposure. The results were once more very similar to the ones obtained with SAFIR, while most of them were close to VULCAN as well. Version 10.0 of VULCAN does not perform thermal analysis. The relationship between temperatures of different parts of the section changes throughout the period of fire exposure. However, these relationships are taken as being constant in the program, and this could be the main cause of the disparities found in the results.

### 3.3 Reinforced concrete beam

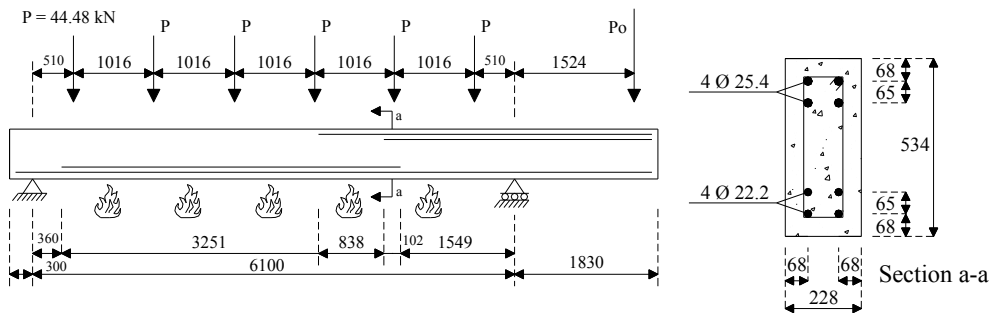


Figure 8 Reinforced concrete beam

Six reinforced concrete beam specimens (one of them schematically shown in Figure 8) were tested by Ellingwood and Lin (1991) and a numerical analysis with VULCAN was carried out for three of them by Cai et al. (2003). The beams analyzed with the present formulation correspond to Beam 3 and Beam 6 from the original work and were exposed to fire on the lateral and lower faces of the cross section, while the cantilever was kept at ambient temperature. Beam 3 was subjected to standard ASTM E119 fire and Beam 6 was exposed to a SDHI (short duration high intensity) fire.

The concrete strength was 29.65 MPa and 34.54 MPa for Beams 3 and 6 respectively, and the reinforcement had 495 MPa yield strength and 187244 MPa elastic modulus at ambient temperature. In the absence of complete information on concrete moisture, its effects were considered in the present work by taking a moisture value of 3% of the concrete weight. The concrete thermal conductivity was taken as the lower limit prescribed by the Eurocode 2 (2004). The load  $P_0$  was taken here as constant and equal to 157.8kN e 111.2kN for Beams 3 and 6, respectively. Cai et al. (2003) employed 20 elements in the numerical analysis, and obtained the cross section temperature distribution with the three-dimensional heat and mass transfer analysis program FPRCBC-T (Huang, et al., 1996). In the present work, 16 elements were used. Figures 9 and 10 show the comparison between the maximum displacements obtained with the test and by numerical analyses. The results are quite reasonable in spite of the uncertainty of some of the parameters involved and the differences in the numerical modeling.

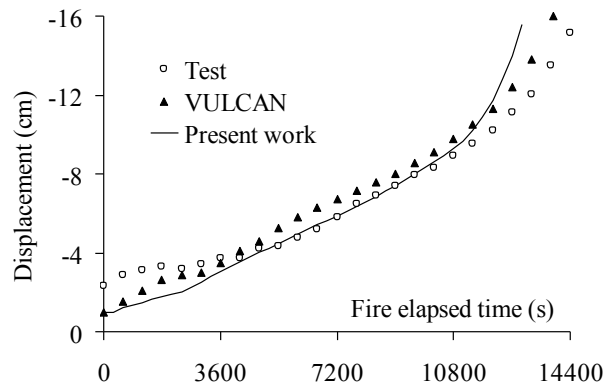


Figure 9 Maximum displacements of the span under fire (Beam 3)

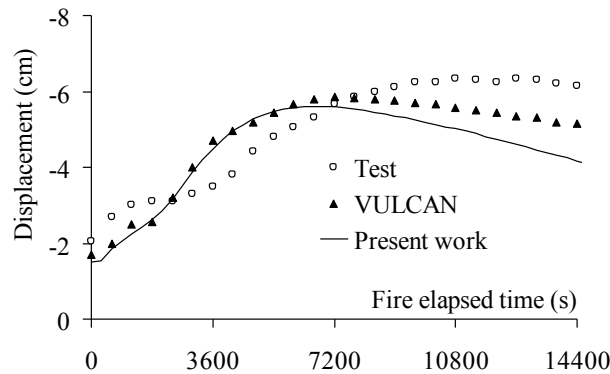


Figure 10 Maximum displacements of the span under fire (Beam 6)

### 3.4 Composite steel and concrete beam

Two composite steel-concrete beams under fire conditions were tested by Wainman and Kirby (1988), and a numerical analysis with VULCAN was carried out by Huang et al. (1999). The beam cross section was composed of a steel section 254x146mm x 43kg/m connected to a reinforced concrete slab 624x130mm with four 8mm bars of longitudinal reinforcement. The steel profile and the reinforcement bars had 255MPa and 600MPa yield strength respectively, and the concrete compressive strength was 30 MPa. The length of both beams was 4530mm and identical concentrated loads were applied at four points on each beam. These loads were 32.47kN and 62.36kN, respectively, for Test 15 and Test 16 [50]. The beams were exposed to standard fire (ISO 834) on their lower face. In the present analysis, material properties were taken from the Eurocode 4 with 8 finite elements with four integration points each. The displacement results (Figure 11) show good agreement between the predictions of the present work and the experimental and numerical values.

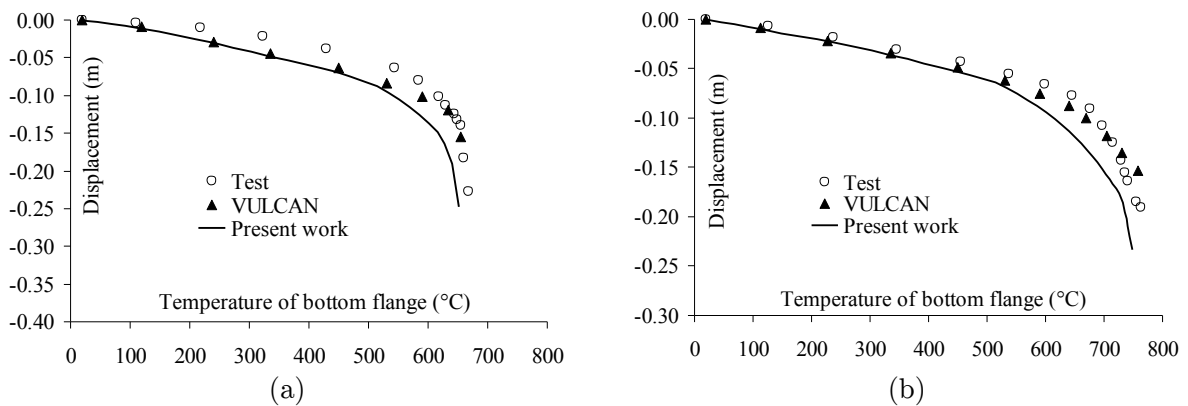


Figure 11 Composite beams under fire: (a) Test 15; (b) Test 16.

### 3.5 3D Steel Frame

Souza Jr. and Creus (2007) present the numerical results of the 3D steel frame depicted in Figure 12, employing SAFIR. All the cross sections are H150x150x7x10mm and the load  $P=250\text{kN}$ . The inferior beams and columns of the structure are subjected to temperature increases. The results for the displacements obtained with the present formulation and their comparison with SAFIR in the directions of loads H1 and H2 are shown in Figure 13.

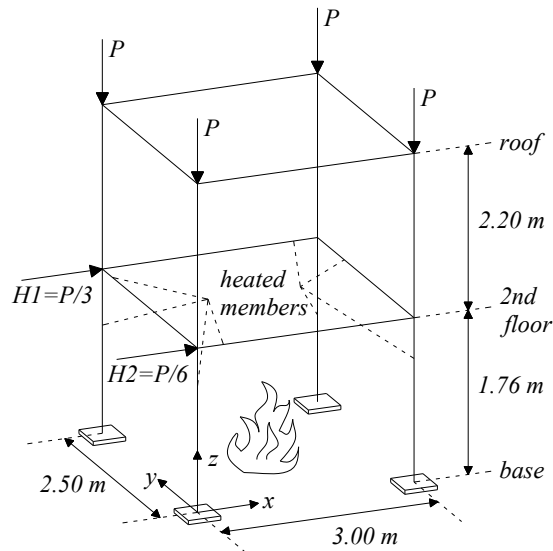


Figure 12 Three-dimensional steel frame

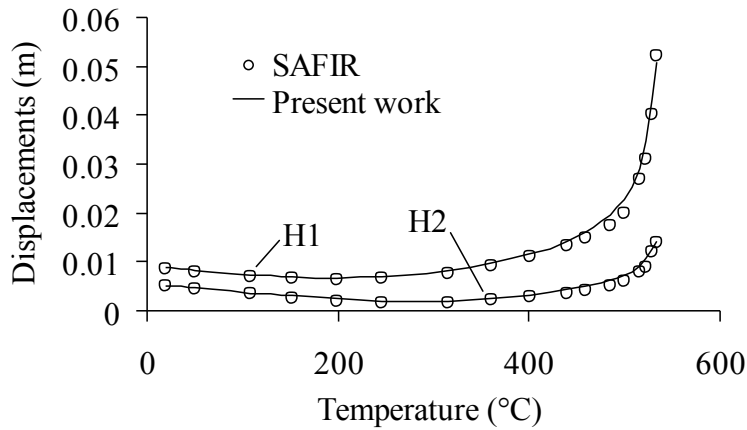


Figure 13 Displacement-temperature results of the 3D steel frame

### 3.6 Discussion

Close agreement between the results obtained in this work and the results of the SAFIR program can be linked with basic assumptions about beam elements considered in both programs. Neither the beam elements presented in this work nor those of the SAFIR program consider shear deformation, but both use corotational formulations. In this work, thermal analysis is performed with the boundary conditions set for the section. For each increment of exposure time to fire, thermomechanical analysis is performed. The SAFIR program performs thermal analysis, stored the temperatures and subsequently performs mechanical analysis (Franssen, 2005).

The VULCAN version 10.0 considers shear deformation and uses a total lagrangian formulation, but the temperatures of parts of the section, in general, are input data, taken as a given percentage of a time-temperature curve defined by the user. Since the increases in temperature between parts of a section are generally not constant during exposure to fire, it is believed that this is the main cause of the small differences between the results obtained in examples 1 and 2.

## 4 CONCLUSIONS

This paper presented the implementation of a finite element for the thermomechanical analysis of 3D framed structures made of bare steel, reinforced concrete or composite steel and concrete, and their application to the analysis of structural behavior in fire conditions. The information presented here can be used and adapted to other models focused on analysis at room temperature. The proposed formulation displayed very good agreement when compared to well-established programs such as VULCAN and SAFIR. Generic cross sections can be considered, providing a valuable tool for the general analysis of framed structures subjected to elevated temperatures.

Future research will focus on the development of models to simulate the behavior of reinforced concrete slabs, and will combine these elements with the beam-column analysis presented here, in order to have a framework for the general analysis of buildings under fire conditions.

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